

AD A102254

DNA 3964F-1-2

THE ROSCOE MANUAL

Volume 1-2: A Simplified ROSCOE Input Scheme

37

(12)

James Baltes
Joel Garbarino
General Research Corporation
P.O. Box 6770
Santa Barbara, California 93111

LEVEL II

29 February 1980

Final Report for Period 9 November 1977-29 February 1980

CONTRACT No. DNA 001-78-C-0002

APPROVED FOR PUBLIC RELEASE;
DISTRIBUTION UNLIMITED.

DTIC
SELECTED
S JUL 31 1981
D

E

THIS WORK SPONSORED BY THE DEFENSE NUCLEAR AGENCY
UNDER RDT&E RMSS CODES B322074464 S99QAXHC06428 H2590D
AND B322075464 S99QAXHC06432 H2590D.

Prepared for
Director
DEFENSE NUCLEAR AGENCY
Washington, D. C. 20305

DTIC FILE COPY

81 7 29 009

Destroy this report when it is no longer needed. Do not return to sender.

PLEASE NOTIFY THE DEFENSE NUCLEAR AGENCY,
ATTN: STTI, WASHINGTON, D.C. 20305, IF
YOUR ADDRESS IS INCORRECT, IF YOU WISH TO
BE DELETED FROM THE DISTRIBUTION LIST, OR
IF THE ADDRESSEE IS NO LONGER EMPLOYED BY
YOUR ORGANIZATION.



UNCLASSIFIED

SECURITY CLASSIFICATION OF THIS PAGE (When Data Entered)

REPORT DOCUMENTATION PAGE		READ INSTRUCTIONS BEFORE COMPLETING FORM	
1. REPORT NUMBER 19 DNA B964F-1-2		2. GOVT ACCESSION NO. AD-A101-254	
4. TITLE (and Subtitle) THE ROSCOE MANUAL Volume 1-2: A Simplified ROSCOE Input Scheme		5. TYPE OF REPORT & PERIOD COVERED Final Report for Period 9 Nov 77-29 Feb 80	
7. AUTHOR(S) 10) James Baltes Joel Garbarino		8. PERFORMING ORGANIZATION NAME AND ADDRESS General Research Corporation P. O. Box 6770 Santa Barbara, California 93111	
11. CONTROLLING OFFICE NAME AND ADDRESS Director Defense Nuclear Agency Washington, D.C. 20305		10. PROGRAM ELEMENT PROJECT TASK AREA & WORK UNIT NUMBERS 15) DNA 111-78-C-0002	
14. MONITORING AGENCY NAME & ADDRESS (if different from Controlling Office) 14) DRC-2R-4-520-VOL-1-2		11. REPORT DATE 29 February 1980	
16. DISTRIBUTION STATEMENT (of this Report) Approved for public release; distribution unlimited.		12. NUMBER OF PAGES 13) 42	
17. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, if different from Report)		15. SECURITY CLASS (of this report) UNCLASSIFIED	
18. SUPPLEMENTARY NOTES This work sponsored by the Defense Nuclear Agency under RDT&E RMSS Codes B322074464 S99QAXHC06428 H2590D and B322075464 S99QAXHC06432 H2590D.		19. KEY WORDS (Continue on reverse side if necessary and identify by block number) Nuclear Effects Computer Program Simulation Radar Optics Satellite Communications	
20. ABSTRACT (Continue on reverse side if necessary and identify by block number) The ROSCOE computer code is designed specifically to be the "laboratory standard" for evaluating nuclear effects on radar and optical sensors. The program provides a means for (1) evaluating radar acquisition, discrimination, and tracking performance in a nuclear environment; (2) determining optical (SWIR) effects; (3) measuring the degradation of microwave satellite communications systems due to transmission through nuclear disturbed regions; (4) estimating various radar and optical propagation error sources; and (5) computing specific phenomenological data. -		16. DECLASSIFICATION DOWNGRADING SCHEDULE	

UNCLASSIFIED

SECURITY CLASSIFICATION OF THIS PAGE/When Data Entered

20. (Continued)

The ROSCOE documentation consists of a number of volumes, including user manuals (Volumes 1 through 3), systems code descriptions (Volumes 4, 20, and 21-1), code validation documents (Volumes 6 and 23), and phenomenology code descriptions (all others). This document has been written as an extension to the user manuals. It describes a simplified input scheme for running a subset of ROSCOE problems. It is intended for the user who only occasionally runs the code or would like to run a small problem.

UNCLASSIFIED

SECURITY CLASSIFICATION OF THIS PAGE/When Data Entered

TABLE OF CONTENTS

<u>Section</u>	<u>Page</u>
LIST OF TABLES	2
1 INTRODUCTION	3
2 DESCRIPTION	4
2.1 Limitations	4
2.2 Input Variables	5
2.3 Example Input Sets	6
2.4 Outputs	8
3 ACCESSING THE INPUT SCHEME	16
3.1 Batch Jobs	16
3.2 Interactive Jobs	16
APPENDIX A: USER REFERENCE TABLES	17

Accession For	
NTIS GRANT	
DTIC TAB	
Unnumbered	
Classification	
By _____	
Distribution/	
Availability Codes	
_____ and/or	
Dist Special	
A	

LIST OF TABLES

<u>Table</u>		<u>Page</u>
1	ROSCOE Tabular Outputs	10
A.1	A Directory of Input Variables	18
A.2	Allowable Unit Names	29
A.3	Position Coordinate Specifications	31
A.4	Sample Control Card Deck for AFWL/NOS/BE1	33
A.5	Sample Procedure Permfile for Interactive Use	34
A.6	Time-Share Inputs	35

1 INTRODUCTION

In the last few years, ROSCOE (Radar and Optical Systems Code with Nuclear Effects) has been expanded to include simulations of satellite communications and optical surveillance systems in a nuclear environment. This expansion has led to considerably more complexity in the input requirements.

While the ROSCOE input scheme was devised to handle these problems (with no additional coding) and to allow the user complete flexibility in structuring scenarios with multiple sensors, objects, and bursts, it takes some time to learn how to use the system. For the user who only occasionally runs the code, or would like to run a small problem, a new input scheme has been built for running a subset of ROSCOE problems with a simple set of inputs.

The next section describes this new input scheme. Example input sets are shown for several different types of problems and the program outputs are briefly discussed. Section 3 describes how to access the new scheme, for both batch and interactive jobs. Finally, to make this paper useful as a reference guide, tables which describe the input options have been placed in Appendix A.

2 DESCRIPTION

The new ROSCOE input scheme consists of a data deck with a pre-selected set of input options, and a data preprocessor program which inserts user-specified values for the options into the data deck. The scheme, in general, does not sacrifice any of ROSCOE's input versatility, since a new data deck with a different set of options can be generated without writing new code.

2.1 LIMITATIONS

With the new scheme, as currently set up, the user can run nuclear burst phenomenology problems alone, or nuclear effects on radar surveillance and tracking of ballistic missiles, satellite communication, or optical surveillance and tracking, subject to these constraints:

- Up to five bursts are allowed, at altitudes up to 400 km -- positions, times, and burst properties are input.
- Only one radar can be simulated in a run -- radar characteristics and location are input.
- Only one object trajectory can be simulated in a run (although multiple objects can be spaced in time on the trajectory) -- launch and impact points, impact time, and reentry angle are input.
- Only one satellite communication system can be simulated in a run (consisting of one ground transmitter, one ground receiver, and one set of satellite-borne equipment which receives and transmits) -- transmitter and receiver characteristics and locations are input.
- Only one optical sensor can be simulated in a run -- sensor characteristics and location are input.
- Run times can be no more than 900 seconds after the last burst.

2.2 INPUT VARIABLES

Input variables in the new scheme are of five types:

- General Inputs. Variables related to a reference location or time.
- Physics Inputs. Variables required to simulate a burst and print physics outputs.
- Radar Inputs. Variables required to simulate radar surveillance or tracking performance.
- Satcom Inputs. Variables required to simulate a satellite communication problem.
- Optics Inputs. Variables required to simulate optical sensor surveillance or tracking performance.

Table A.1 is a directory of input variables, divided into the five types described above with notes to indicate the options available. For each variable, the table gives its name, the number of values to be supplied (more than one if the variable is a vector), a definition of the variable including default units of measure, the default values that will be assumed if you do not input the variable, and whether a unit name is allowed for the variable. (Table A.2 shows the allowable unit names.) It is important to note the default units given. If you input values without unit names (for those variables allowing unit names), the default units are assumed. Note that the default values listed in Table A.2 are given in their customary units, which are not always the same as the internal default units.

To run a case, follow the instructions given in Table A.1, and input those variables you wish to change in the form: variable = value unit, variable = value unit, etc. End the input string with the command RUN following the last variable input. For vectors, the format may be: vector = value unit, value unit, etc., or vector(index) = value unit, value unit, etc. In the first case, the values are assigned to vector(1),

vector(2), etc.; in the second case, values are assigned to vector(index), vector(index + 1), etc. This free format is essentially compatible with the Fortran NAMELIST input scheme.

Note that positions can be specified by geographical coordinates (GEOGR), or by Cartesian (LOCXYZ) or range-azimuth-elevation (RADAR) coordinates relative to a reference location. The order of entry, orientation, and units for these specifications are given in Table A.3 and Fig. A.1.

2.3 EXAMPLE INPUT SETS

2.3.1 Physics Problem

To run a simple physics problem consisting of a single burst with the default characteristics and these assumptions:

- Burst time = 0 s
- Yield = 10 kT
- Altitude = 40 km
- Output every 20 s until 120 s after burst

input:

```
TSTOP = 120, OTIME = 0, OTINT = 20, BTIME1 = 0,  
BP0S1(3) = 40, YIELD1 = 10 KT, RUN
```

2.3.2 Radar Problem

To run a radar surveillance problem, where:

- There is a single burst with the above properties.
- The radar is at the center of a local Cartesian coordinate system (directly under the burst).
- The radar is of the type described by the default parameters.
- The object being viewed has a -30° reentry angle and is aimed at the radar.

- The object is at 100 km altitude at time = 0 when the burst occurs.
- Radar measurements are made once every second for 20 s.

input:

TST \emptyset P = 20, BTIME1 = 0, BP \emptyset S1(3) = 40, YIELD1 = 10 KT,
 \emptyset BTAG = \emptyset BJECT-1, \emptyset BTIM = 0, \emptyset BP \emptyset S(2) = 173,
100 KM, \emptyset BVEL(3) = -30, RADAR = REFER, RUN

2.3.3 Satcom Problem

To run a satellite communication problem, where:

- The ground transmitter and receiver are together, directly beneath a satellite at synchronous altitude (the default condition)
- The default link inputs are assumed
- The default nuclear burst (1 MT at 200 km altitude) occurs 10 s after the first communication
- The burst is displaced 200 km horizontally from the line of sight
- Communication calculations are made every 20 s, from 0 s to 100 s

input:

TST \emptyset P = 100, BTIME1 = 10, BP \emptyset S1(2) = 200, CTIME = 0,
CTINT = 20, RUN

2.3.4 Optics Problem

To run an optical sensor surveillance problem, where:

- There is a single burst of 10 kt at 40 km altitude
- The sensor is at synchronous altitude
- The sensor is pointed at the burst

- The sensor is of the type described by the default parameters
- Sensor calculations are made at only one time (0 s)

input:

```
TSTOP = 1, BTIME1 = 0, BP0S1(3) = 40, YIELD1 = 10 KT,  
0BTAG = REF-0BJECT, 0TYPE = SURVEILLANCE, 0L00K = 0,  
REFPT(1) = 40, 0PTICS = REFER, RUN
```

2.4 OUTPUTS

The outputs produced by the ROSCOE code using the new input scheme are described in this section. Two types of outputs may be produced: (1) printer plots, and (2) tabular outputs.

2.4.1 Printer Plots

When a high-altitude (>90 km) burst is simulated, the code produces a series of printer plots at times specified by the 0TIME, 0TINT input variables. The plots consist of a picture of the fireball and beta tube region and contour plots of mass density, electron density, and striation fraction in the high-altitude grid.

The contour plots of mass density and electron density represent vertical cross sections through the burst point in the (magnetic) north-south direction, viewed looking eastward. The contour plots of the striation fraction are cross sections normal to the earth's magnetic field, viewed looking down the field lines.

The plots are produced as they are computed internally, and thus will appear before the tabular output described below.

In addition, contour plots of the relative radiance at the focal plane of the sensor can be generated when the optics code is used. These plots are generated when the optics calculation type, OCAL, is set to FOV.

2.4.2 Tabular Outputs

There are seven phenomenology lists, five radar lists, two satellite-communication lists, and three optics lists that may be output at the conclusion of the run, depending on the type of simulation performed.

The phenomenology lists include: burst parameters, common fireball parameters (fireball set 1), two additional low-altitude fireball parameter lists (fireball set 2 and fireball set 3), additional high-altitude fireball parameters (fireball set 4), contained debris region parameters, and beta tube parameters.

The radar lists include: trajectory output, track measurement errors, track filter output, and two lists of propagation errors.

The satcom lists include: propagation and probability-of-error data, and satellite position coordinates with respect to the ground-terminal positions.

The optics output lists include: angle and signal-strength measurements for an optical tracking sensor application, the radiance along each path treated within the field-of-view, and the data stream output produced by a scanning sensor.

Table 1 shows a small sample of each type of output. Some of the column headings are self-explanatory, while others require additional comment.

TABLE 1
ROSCOE TABULAR OUTPUTS

PHENOMENOLOGY OUTPUTS

BURST PARAMETERS		
CF SEC	TIME OF BURST (SECS)	BURST ENERGY (TERJ)
		BURST PT. ALTITUDE KM
1030.000	4116.23	20921.23
1060.000	4116.23	20921.23
		250.000
		250.000
		5045E-13
		5411E-12
		67.922
		13055.556
		37.579
		3937.792
		212.560
		0.000
		0.000

NOTE: Columns 9 and 10: The outputs "time to reach 3000 K and 2000 K" are used only for low-altitude (<90 km) fireball chemistry calculations.

FIREBALL SET-1		
CF SEC	TIME OF BURST NUMBER	VERTICAL RADIUS KM
		REFINED ALTITUDE KM
1030.000	1	280.513
1060.000	1	280.513
		280.073
		421.560
		475.551
		1.671
		1.471
		0.000
		0.000

FIREBALL SET-2		
CF SEC	TIME OF BURST NUMBER	MAXIMUM ALTITUDE KM
		VERTICAL ROTATION DEG
1030.000	1	164.274
1060.000	1	195.713
		690.661
		901.723
		5.540
		713
		0.000
		280.533
		269.073
		426.172
		0.000
		0.000

NOTE: Column 6: Axis rotation is measured +CCW from North. Column 10: The characteristic time is the approximate time this fireball has merged with another (used only for low-altitude fireballs).

Table 1 (Continued)

FIREBALL SET-3						
TIME OF INPUT SEC	FIREBALL NUMBER	Y ₀ C ₀ (M)	Z ₀ C ₀ (M)	Y ₀ C ₀ (M)	Z ₀ C ₀ (M)	Y ₀ C ₀ (M)
95.000	1	-0.115E+09	-0.451E+09	0.421E+09	0.500	0.000
96.000	1	-0.115E+09	-0.451E+09	0.421E+09	0.411	0.000

NOTE: Column 6: The Oval of Cassini parameter describes the shape of a low-altitude fireball. A value of 1.0 or greater means the fireball has formed a torus. Columns 9 and 10: The fireball kind can take values from 1 to 5, where: 1 = spheroid, 2 = skewed spheroid, 3 = torus, 4 = inactive radiation-merged fireball, 5 = inactive hydromerged fireball.

FIREBALL SET-4						
TIME OF INPUT SEC	FIREBALL NUMBER	Y ₀ C ₀ (M)	Z ₀ C ₀ (M)	Y ₀ C ₀ (M)	Z ₀ C ₀ (M)	Y ₀ C ₀ (M)
1630.000	1	-0.491E+09	-0.501E+09	0.501E+09	3	3
1640.000	1	-0.492E+09	-0.501E+09	0.501E+09	3	3

NOTE: Columns 6 to 8: The grid cell indices refer to the grid cell in which the fireball center is located.

DEBRIS PARAMETERS						
TIME OF INPUT SEC	FIREBALL NUMBER	DEBRIS NUMBER	TOTAL FLIGHT TIME(S)	DEBRIS ALTITUDE KM	VERTICAL RADIUS KM	LEWIS DISTANCE KM
95.000	1	1	0.834E+20	0.820	0.053	0.000
96.000	1	1	0.834E+20	0.804	0.070	0.000

NOTE: Column 8: The debris distribution parameter describes the rate of fall-off of the bolt source strength from the tube boundary (see RANC IV).

Table 1 (Continued)

BETA TUBE PARAMETERS						KINK-RADIUS			KINK-RADIUS		
TYPE	FLUTERALL	RETARDATION	INITIAL	KINK-ANGLE	KINK-PURST	AT 85KM	AT 85KM	AT 85KM	AT 60KM	AT 60KM	AT 60KM
OF INPUT	INCLINA	SHAPE	DIP ANGLE	FROM	MIN. DISTANCE	KM	KM	KM	KM	KM	KM
SEC			0.0								
1630.000	1	KINK	76.506	76.674	40.259	203.619	207.111	202.547	206.309		
1650.000	1	KINK	76.506	77.871	40.326	201.169	205.439	199.915	204.644		

NOTE: Column 3: The beta tube shape is either "STRAIGHT" or "KINK". Column 6: The kink-burst distance is the distance from the sub-burst point at 85 km to the center of the beta tube at 85 km.

RADAR OUTPUTS

RADAR OUTPUT						TIME-VELOCITY			NUMBER OF TARGETS		
TYPE	TYPE	POSITION	DATA FOR	OBJECT AT	SPECIFIED	TIME	VELOCITY	SIGNAL TO	NOISE (DB)	NUMBER OF	TARGETS
OF EVENT	OF OUTPUT	ATTITUDE	ANGLE	AZIMUTH	ELEVATION	SEC	M	SEC	SEC	SEC	SEC
SEARCH	1519.497	910.669.377	3236.670.933	81.274	2.722	6226.094	19.879	1			
VERIFY	1519.597	910.669.377	3236.670.933	81.274	2.722	6226.759	20.575	1			

NOTE: Column 1: The event type is either "SEARCH", "VERIFY", "TRACK IN" (for track initiation), or "TRACK". Columns 3 to 7: The position and velocity data given here are the actual values. Column 9: The number of targets can be zero if the target has been lost, one if a single target has been located, or more than one if multipath effects occur.

TRACK "F" SURVEY POINTS

TRACK "F" SURVEY POINTS						MEASURED			MEASURED		
TYPE	POSITION	PROJECTED	MEASURED	MEASURED	MEASURED	RANGE	PAULIN	RANGE	PAULIN	RANGE	PAULIN
OF OUTPUT	WAVE	AZIMUTH	ELAT.	ELAT.	ELAT.	DEG	DEG	DEG	DEG	DEG	DEG
1539.497	129.670.933	81.274	2.722	3236.610.359	81.182	2.604	140.574	0.000	0.000	0.000	0.000
1539.597	129.680.508	81.274	2.722	3236.611.211	81.182	2.604	130.403	0.000	0.000	0.000	0.000

NOTE: Columns 2 to 4 and 5 to 7: The predicted position is either equivalent to the actual position for search pulses or is the position predicted by the track filter once track has been initialized. The measured coordinates are those generated during the current look and include all refraction and radar measurement errors.

Table 1 (Continued)

TRACKING FEATURES									
TIME	POSITION IN ALTIM V	PREDICTED POSITION IN ALTIM V	POSITION IN ALTIM V	VELOCITY IN PRED TR V	VELOCITY IN PRED TR V	APPARENT HEIGHT	TARGET AZIMUTH DEG	POSITION ELEVATION VEG	
1616.997	112.146	114.9.6	117.792	1718.544	6084.306	-2792.826	3193.76.724	80.954	
1617.097	112.660	113.467	1183.290	1151.001	2212.609	366.810	3167.482.417	80.918	
								3.250	
								3.284	

PROPAGATION INPUT - I									
TIME	ABSORPTION FROM ALL SOURCES	REFLECTION FROM ALL SOURCES	INCIDENT POWER	CLUTTER TO-NOISE RATIO (DB)	DISPERSIVE LOSS	Faraday ROTATION LOSS	FAILURE MODE	FAILURE MODE	
1509.497	0.000	7.206	0.000	0.2455.009	0.000	1.000	NO FAILURE	NO FAILURE	
1509.597	0.000	7.583	0.000	0.2455.009	0.000	1.000	NO FAILURE	NO FAILURE	

NOTE:	Column 9: The failure mode flag can have the following messages:
NO FAILURE	S/N received is above threshold
RANGE	The radar is range (power) limited for this target
ABSORPTION	The absorption due to all sources has reduced the S/N below threshold
ABS-NOISE	The combination of absorption and fireball noise has reduced the S/N below threshold
TOTAL	The combination of absorption, noise, dispersion, and Faraday rotation has dropped the S/N below threshold
LOW SIGNAL	The combination of the above effects and refraction or clutter has dropped the S/N below threshold
NO TARGET	There are no targets within the range gate and 3 dB beamwidth

Table 1 (Continued)

PROPAGATION OUTPUT-2					
TIME OF OUTPUT SEC	RANGE M	BIAS DEG	REFRACTION AZIMUTH DEG	ERRATUS ELEVATION DEG	RANDOM RANGE M
1500.497	0.000	0.000	0.000	0.000	0.000
1501.497	0.000	0.000	0.000	0.000	0.000
SATCOM OUTPUTS					

COMMUNICATIONS OUTPUT-1

TYPE OF OUTPUT SEC	TIME OF OUTPUT SEC	UPLINK LOSS FACTOR	UPLINK SCINT	DOWNLINK LOSS FACTOR	DOWNLINK SCINT
CCW-EFCVO FCP-BTCVO	1612.000 1622.000	1.001 22.403	0° 9526.6	1.615 60.143	0° 10230.

NOTE: Columns 3 to 6: The uplink and downlink loss factors are the losses due to absorption from all sources (dimensionless). The uplink and downlink scintillation values refer to the standard deviation in phase due to scintillation effects in radians.

COMMUNICATIONS OUTPUT-2

TIME OF OUTPUT SEC	SE TELITE RANGE M	COMBI. RPT AZIMUTH DEG	TRANSMIT ELEVATION DEG	SATELLITE RANGE M	COMBI. RPT AZIMUTH DEG
1612.000	1506.105	0.00.092	74.592	1306.105	-0.00.892
1622.000	1291.663	0.00.922	77.716	1291.663	-0.00.922

Table 1 (Continued)
OPTICS OUTPUT

NOTE: The radiance in Column 5 is the integrated radiance along the path (described by the azimuth and elevation off-boresight) due to all emission and scattering sources. The integrated radiance in Column 6 is just radiance integrated over all band intervals and the *size* is due to structure (Column 7) is the deviation in the integrated radiance due to striated (or structured) regions along the path.

3 ACCESSING THE INPUT SCHEME

3.1 BATCH JOBS

To access and use the new input scheme in the batch mode (i.e., by submitting a card input deck over the counter or through a remote terminal), use a deck setup such as that shown in Table A.4.

Note that an optional card may precede the data cards, directing the input program to print each default card changed, followed by the new card which replaces it.

3.2 INTERACTIVE JOBS

To access and use the new input scheme using the time-share system follow these steps (also shown in Table A.6). (First, you must have a procedure permfile containing a small CYBER control language "PRØC" and a set of control cards. A sample procedure permfile is shown in Table A.5)

- Step 1. Access your procedure file with the ATTACH statement.
- Step 2. Execute the ROSCOE time-share program by typing RØSCØTS.
- Step 3. Type your inputs, in response to the program's message "INPUTS?". The program then processes the inputs; that is, inserts them into the standard deck and checks for errors. If errors occur, the program prints them and asks you to input a revised list by again asking "INPUTS?". When no errors occur, terminate RØSCØTS by typing "RUN". The job file is then automatically placed in the input queue, and control returns to the INTERCOM system. You can check that your job has been accepted by typing a FIND, nnn command, where nnn is the first 1-5 characters of the job name (first parameter on your first control card).

APPENDIX A

USER REFERENCE TABLES

DICTIONARY OF INPUT VARIABLES

TABLE A.1
FACTORY OF INPUT VARIABLES

A DIRECTORY OF INPUT VARIABLES			02/04/80	10.36.16.	PAGE 1
INPUT VARIABLE	NO.	VALUES	DEFAULT	VALUES	UNIT-NAME ALLOWED
DISCRIMIN.					

卷之三

A. GENERAL INPUTS

- THE DEFECT STALLS AND SET SO THAT THE CODE PROCESSES THE STEP EASILY FIRST AND THIS PRECLUDES THE STEP FROM BEING EXECUTED.
- TO HIGH PHYSICAL MEMORY, SWAP, OR OPTICS PROBLEMS CREATE THE EVENT TIMES DESCRIBED BELOW TO OCCUR BEFORE THE STEP TIME.

TEM STUP TIME (LEFAULT UNIT IS SEC)
REFERENCE POSITION ALTITUDE (M)
REFERENCE POSITION LATITUDE (DEGREES)
REFERENCE POSITION LONGITUDE (DEGREES)

REFERENCE POSITION LAST LOGGED (FC)
REFERENCE POSITION, NORTH LATITUDE (FC)
TO SUPPRESS EVENT LIST CUTOFF, SET THIS LINE TO ZERO. A SINGLE
PORT LINE IS PRINTED FOR EACH EVENT PRECESSION
PORT LINE SET TO HIGH ALTITUDE (MOVE SC_KP1 GRIC (KAC))

THE ECH ALTECH CANTER (2000 ft. 200 ft. 3-MAINTIME)
LITY MAINTEN FCH. EUSC CANTER (1500 ft. 23 ft. 4-
4500 ft. 22 ft.)

FOR LONGITUDINAL ORGANIZATION—FLAT TO DIFFUSE— $\alpha=0.1$
FOR LONGITUDINAL ORGANIZATION—STATISTICAL CLOUDS IN UNLATTICED
CLOUDS— $\alpha=0.1$ FOR LONGITUDINAL CLOUDS, CLOUDWISE

8. PHYSICS LOCUS INPUTS

--10. RUN A PHYSICS PROBLEM. INPUT THE BURNT TIMES
1. INPUT BURNT TIME, ETC. INPUT CCCM PRIGR TC TNGR
--FOR EXAMPLE--BY INPUTTING P1=1.0, T1=20.
--THE CYCLE SIMULATES A SINGLE BLAST FOR TIME
SOL TO 120 SEC AFTER LAST.

DIRECTORY OF INPUT VARIABLES
Table A.1 (Continued)

INPUT NO. VARIABLE DESCRIPTION
 VALUES

1. RUN CONTROL

--TIL COCE PHOVILES ONLY BURST PARAMETER OUTPUT BY
LEFFLT. TO GET TIME STEP FOR ALL TIME, CLEARS
PROBLEMS AT HIGHLEVEL INTERVALS. INPUT TIME1 AND
TIME2.

--FOR EXAMPLE--BY INPUTTING RTIME1=0, SEC, RTIME2=1, SEC
CRLT=30, SEC, THE COCE WILL OUTPUT PHYSICS DATA
STARTING AT 1 SEC AFTER BURST AND CONTINUE EVERY 30
SEC TIL TIME2. TIL THE STEP TIME IS REACHED.

RTIME 1 0TIME = PHYSICS OUTPUT TIME (LEFFLT UNIT IS SEC)
 OTINT 1 CRINT = PHYSICS OUTPUT DATA TIME INTERVAL (LEFFLT UNIT IS SEC)

2. BURST DATA

--UP TO FIVE BURSTS ARE ALLOWED.
--THE USER CAN CHOOSE TO TREAT BURST COORDINATES IN
GEOGRAPHICAL COORDINATES (GLOB) OR COORDINATES
RELATIVE TO THE REFERENCE LOCATION IN SECTION A. ABCVE
LOCXYZ OR HAZAR.

--FOR EXAMPLE--THE USER CAN INPUT BPOS1=0,0,50,50,LOCXYZ
AND BURST 1 IS SPECIFIC TO A CARTESIAN EAST-NORTH-UP
(XYZ) COORDINATE SYSTEM.

BTIME1	1	BTIME1 = BURST TIME FOR BURST 1 (LEFFLT UNIT IS SEC)	99999.	SEC
BTIME2	1	BTIME2 = BURST TIME FOR BURST 2 (LEFFLT UNIT IS SEC)	99999.	SEC
BTIME3	1	BTIME3 = BURST TIME FOR BURST 3 (LEFFLT UNIT IS SEC)	99999.	SEC
BTIME4	1	BTIME4 = BURST TIME FOR BURST 4 (LEFFLT UNIT IS SEC)	99999.	SEC
BTIME5	1	BTIME5 = BURST TIME FOR BURST 5 (LEFFLT UNIT IS SEC)	99999.	SEC
BTPOS1	4	BTPOS1(1-4) = BURST COORDINATES FOR BURST 1. COORDINATE1(1)=AST LONGITUDE (DEG), X-COORDINATE, ON HAZAR(LM). COORDINATE2(2)=AST LATITUDE (DEG), Y-COORDINATE, ON ASPECT-CLM FRCY EAST(3)= LATITUDE(4)=AST LATITUDE (DEG). ON ELEVATION(CL4).	0.	NO
BTPOS2	4	BTPOS2(1-4) = HAZAR COORDINATES FOR BURST 2 (SEE BPOS1 DESCRIPTION)	0.	NO
BTPOS3	4	BTPOS3(1-4) = BURST COORDINATES FOR BURST 3 (SEE BPOS1 DESCRIPTION)	0.	NO
BTPOS4	4	BTPOS4(1-4) = BURST COORDINATES FOR BURST 4 (SEE BPOS1 DESCRIPTION)	0.	NO
BTPOS5	4	BTPOS5(1-4) = BURST COORDINATES FOR BURST 5 (SEE BPOS1 DESCRIPTION)	0.	NO
			LOCKXYZ	NO

DIRECTORY OF INPUT VARIABLES
Table A.1 (Continued)

INPUT VARIABLE	NC. VALUES	DISCRIMIN

02/04/60 10:36:18.
REFLUT
VALUES
0.
200.
LOCKTZ

		PAGE 3 UNIT-NAME ALLOCED
YIELD1	YIELD OF BURST 1 (LEFFAL UNIT IS MT)	WT
YIELD2	YIELD OF BURST 2 (LEFFAL UNIT IS MT)	WT
YIELD3	YIELD OF BURST 3 (LEFFAL UNIT IS MT)	WT
YIELD4	YIELD OF BURST 4 (LEFFAL UNIT IS MT)	WT
YIELD5	YIELD OF BURST 5 (LEFFAL UNIT IS MT)	WT
FFRAC1	FRSSION FRACTION OF BURST 1	WT
FFRAC2	FRSSION FRACTION OF BURST 2	WT
FFRAC3	FRSSION FRACTION OF BURST 3	WT
FFRAC4	FRSSION FRACTION OF BURST 4	WT
FFRAC5	FRSSION FRACTION OF BURST 5	WT
XFRA1	X-RAY FRACTION OF BURST 1	WT
XFRA2	X-RAY FRACTION OF BURST 2	WT
XFRA3	X-RAY FRACTION OF BURST 3	WT
XFRA4	X-RAY FRACTION OF BURST 4	WT
XFRA5	X-RAY FRACTION OF BURST 5	WT
GFRA1	GFRA1 FRACTION OF BURST 1	WT
GFRA2	GFRA1 FRACTION OF BURST 2	WT
GFRA3	GFRA1 FRACTION OF BURST 3	WT
GFRA4	GFRA1 FRACTION OF BURST 4	WT
GFRA5	GFRA1 FRACTION OF BURST 5	WT
WMASS1	WEAPCN MASS OF BURST 1 (ICEFAULT UNIT IS GM)	GM
WMASS2	WEAPCN MASS OF BURST 2 (ICEFAULT UNIT IS GM)	GM
WMASS3	WEAPCN MASS OF BURST 3 (ICEFAULT UNIT IS GM)	GM
WMASS4	WEAPCN MASS OF BURST 4 (ICEFAULT UNIT IS GM)	GM
WMASS5	WEAPCN MASS OF BURST 5 (ICEFAULT UNIT IS GM)	GM
XITEM1	X-RAY TEMPERATURE (REV) OF BURST 1 -- E.G. 0.5 (0.5,1.0, 2.0)	WT

3. *EARTH DATA
---TC FIVE STATIC TRIPLE CNT. PL ENTERED.
---X-RAY TEMP MATCHES ARE LIMITED TO THE THREE VALUES
LISTED.

YIELD1 = YIELD OF BURST 1 (LEFFAL UNIT IS MT)
YIELD2 = YIELD OF BURST 2 (LEFFAL UNIT IS MT)
YIELD3 = YIELD OF BURST 3 (LEFFAL UNIT IS MT)
YIELD4 = YIELD OF BURST 4 (LEFFAL UNIT IS MT)
YIELD5 = YIELD OF BURST 5 (LEFFAL UNIT IS MT)
FFRAC1 = FRSSION FRACTION OF BURST 1
FFRAC2 = FRSSION FRACTION OF BURST 2
FFRAC3 = FRSSION FRACTION OF BURST 3
FFRAC4 = FRSSION FRACTION OF BURST 4
FFRAC5 = FRSSION FRACTION OF BURST 5
XFRA1 = X-RAY FRACTION OF BURST 1
XFRA2 = X-RAY FRACTION OF BURST 2
XFRA3 = X-RAY FRACTION OF BURST 3
XFRA4 = X-RAY FRACTION OF BURST 4
XFRA5 = X-RAY FRACTION OF BURST 5
GFRA1 = GFRA1 FRACTION OF BURST 1
GFRA2 = GFRA1 FRACTION OF BURST 2
GFRA3 = GFRA1 FRACTION OF BURST 3
GFRA4 = GFRA1 FRACTION OF BURST 4
GFRA5 = GFRA1 FRACTION OF BURST 5
WMASS1 = WEAPCN MASS OF BURST 1 (ICEFAULT UNIT IS GM)
WMASS2 = WEAPCN MASS OF BURST 2 (ICEFAULT UNIT IS GM)
WMASS3 = WEAPCN MASS OF BURST 3 (ICEFAULT UNIT IS GM)
WMASS4 = WEAPCN MASS OF BURST 4 (ICEFAULT UNIT IS GM)
WMASS5 = WEAPCN MASS OF BURST 5 (ICEFAULT UNIT IS GM)
XITEM1 = X-RAY TEMPERATURE (REV) OF BURST 1 -- E.G. 0.5 (0.5,1.0, 2.0)
XITEM2 = X-RAY TEMPERATURE (REV) OF BURST 1 -- E.G. 0.5 (0.5,1.0, 2.0)

DIRECTORY OF INPUT VARIABLES

Table A.1 (Continued)

1st UT

NC

VARIABLE

VALUES

XITEM2	1	XITEM2 = X-HAT TEMPLATURE (NEU) OF BURST 2 --	(0.5,1.0, 0.0)	1.0			NC
		IMPLST BE TYPE1 LITERALLY -- E.G. C.5 NOT .5)					
XITEM3	1	XITEM3 = X-HAT TEMPLATURE (NEU) OF BURST 3 -- (0.5,1.0, 0.0)		1.0			NC
		IMPLST BE TYPE2 LITERALLY -- E.G. C.5 NOT .5)					
XITEM4	1	XITEM4 = X-HAT TEMPLATURE (NEU) OF BURST 4 -- (0.5,1.0, 0.0)		1.0			NO
		IMPLST BE TYPE3 LITERALLY -- E.G. C.5 NOT .5)					
XITEM5	1	XITEM5 = X-HAT TEMPLATURE (NEU) OF BURST 5 -- (0.5,1.0, 0.0)		1.0			NO
		IMPLST BE TYPE4 LITERALLY -- E.G. C.5 NOT .5)					

C. RADAR CODE INPUTS

--TC RUN A RADAR PROBLEM THE USER MUST FIRST SET
RADARREFERER

--RADAR FACILITY CAN THEN BE SET UP IN TWO WAYS--
(1) AN OBJECT POSITION, VELOCITY AND TIME CAN BE INPUT
(SEE OBJECT DATA--STATIC VECTOR INPUT SECTION). ALL THE
FIRST RADAR LOCK WILL BE INITIATED AT THE OBJECT TIME
SPECIFIED. OR
(2) THE USER SPECIFIES AN OBJECT TRAJECTORY (SEE OBJECT
DATA--TRAJECTORY INPUT) AND THE FIRST RADAR LOCK IS
ESTABLISHED AT THE OBJECT ENTERS THE RADAR FGV.

1. RUN CONTROL

--IT DEFERENCE RADAR LOCKS OR CLOSED LCCP TRACK
IS SELECTED BY SETTING (KFLAG).
--SUBSEQUENT LOCKS ARE CREATED INTERNALLY EVERY (CT) SEC.

RADAR	1	KRADAR = FLAG FOR INITIALIZING RADAR PHASE (SET RADARREFERER FOR RADAR CALCULATIONS)	NO
KFLAG	1	KFLAG = FLAG FOR CLOSED LCCP TRACK (0=TRACK, 1=NO TRACK/SEARCH ONLY)	NO
DT	1	CT = RADAR LOCK (IN TRACK) INTERVAL (CTFACTOR UNIT IS SEC)	YES

2. RADAR DATA

--THE RADAR LOCATION CAN BE INPUT IN GEOGRAPHICAL
COORDINATES (LAT/LON) OR RELATIVE TO THE REFERENCE
LOCATION IN SECTION A. ABOVE.
--FOR EXAMPLE-- THE DEFAULT VALUES FOR RADPOS PLACE THE
RADAR AT THE ORIGIN OF THE CARTESIAN EAST-NORTH-UP

DIRECTORY OF USEFUL VARIETIES

Table A.1 (Continued)

VARIABLE	VALUES	DESCRIPTION
		(XXX) COORDINATES PERTICULAR EARLIER.
RDPOS	4	$RCFCS(1-4) = RAYAR POSITION. (RFLG<11-3>=POSITION CCR& RCFCS(1-4)= CURN TYPE<11-3>=LOCXYZ. CH RAYAR) DISTANCES ARE IN MM.$
		ANGLES IN DEG.
BORT	2	$RCFCS(1-4) = RAYAR EARTH POSITION (AZIMUTH CCW FROM EAST, AND ELEVATION IN DEG)$
	1	$RFLG = RAYAR FREQUENCY (CEFAULT UNIT IS Hz)$
	1	$RSQW = RAYAR RAYAR SNR IN 1-5-5-5 TARGET UNITS AVAILABLE AS UNIT NAMES ARE RSDM$
	1	$EEARM = RAYAR EARTH POSITION CM THE CEFAULT UNIT IS CM/SEC/M$
	1	$EEARM = RAYAR EARTH POSITION (CEFAULT UNIT IS RADIANS)$
SEAM	1	$EWAX = MAXIMUM ELEVATION (EJECTION FAULT) FAULT UNIT IS CM)$
RWAY	1	$EWAX = MAXIMUM ELEVATION ANGLE (EJECTION FAULT UNIT IS RADIANS)$
	1	$EWIN = MINIMUM ELEVATION ANGLE (EJECTION FAULT UNIT IS RADIANS)$
SNDV	1	$SNDV = SYNTHETIC ACOUSTIC FOR VERIFICATION. (CEFAULT UNIT IS RADIAN/INCH/SQUARE)$
	1	$SNDV = SYNTHETIC ACOUSTIC FOR TRACK (CEFAULT UNIT IS RADIAN/INCH/SQUARE)$
SUMIT	1	$SMIN = SYNTHETIC RESPONSE FOR TRACK (CEFAULT UNIT IS Hz)$
BRANCH	2	$BRANCH = ACISE BRANCH -->AFCIC 1.0/FULSE LENGTH (CEFAULT UNIT IS Hz)$
BATDOS	1	$BATC = SIGNAL GAIN/10^-4-->AFCIC. ENHANCED CEFAULT UNIT IS Hz$
FPC	1	$FPC = PULSE COMPRESSION HATAC$
RGATE	1	$RGATE = HATAC HATAC GATE FOR TRACK INITIATION. (CEFAULT UNIT IS CM)$
TGATE	1	$TGATE = RANGE GATE FAIRWATER FOR TRACK (CEFAULT UNIT IS CM)$
FERR	3	$FERR(1-3) = FIXED FOR TRACK UP RANGE HATAC MEASUREMENT ERRORS II. RAE COORD (SIGMA(1)*2*FERR(1)*2*STDEV(1)*2*STDEV(1)*2/STK)$
		(CEFAULT UNITS ARE CM, RAD, RAD)
SNERR	3	$SNERR(1-3) = S/N DEPENDENT FOR TRACK HATAC MEASUREMENT ERRORS IN RAD CLOCK. (SIGMA(1)*2*FERR(1)*2*STDEV(1)*2*STDEV(1)*2/STK)$
		(CEFAULT UNITS ARE CM, RAD, RAD)

2000-2001 STATE

DIRECTORY OF INPUT VARIABLES

Table A.1 (Continued)
INPUT NC.

VARIABLE VALUES

DESCRIPTION

02/04/80 10.36.18.

PAGE 6

INPUT-NAME
ALLO-EC

			DEFAULT VALUES	PAGE
BETA	1	BETA = HY ELLIPTICAL COEFFICIENT (ALWAYS INPUT IN FSF)		6
RCS	1	RCS = RADAR CROSS SECTION OF HY (REFALY UNIT IS CPS)		
		- STATE VECTOR INPUT		
OBJREF	1	OBJREF = FLAG TO SELECT OBJECT COORDINATES AS THE BURN IN PCU (STL = REFERENCE TIME FOR THE BURN)	REFEKF	NO
OBJTAG	1	OBJTAG = FLAG TO SELECT A POLAR COORDINATE SYSTEM INPUT SET = ZERCS (STL = TARGET-EFFECT-1 FOR RACAR, A/C CETA-EFFECT-OBJECT FOR CEFICS)	REF-OBJECT	NO
OBJTIM	1	OBJTIM = OBJECT TIME (REFALY UNIT IS SEC)	99999.	YES
OBJPOS	4	OBJPOS(1-4) = OBJECT POSITION (PCUS(1-4)=EPCRS(1-4)= CCOC TYPE=CECR,LOCXYZ, CH RADI, INSTANCES IN KM. ANGLES IN CEG)	0. 50. 50. 70.	NC NO NO NO
OBJVEL	3	OBJVEL(1-3) = OBJECT VELOCITY IN POLAR CRCH (MAGNITUDE IN KM/S, HEADING COUNTERCLOCKWISE FROM LOCAL GEOGRAPHICAL EAST IN CEG, ELEVATION ABOVE LOCAL HORIZONTAL IN DEG)	-90. -45.	NO NO
		- TRAJECTORY INPUT		
NOBJ	1	NOBJ = NUMBER OF OBJECTS ON THE TRAJECTORY	0.	NO
OBJPOS	4	BCEPOS(1-4) = LAUNCH (IN BURN) POSITION (ACPS(1-3)= POSITION CCOC, EPCRS(4)=ECCH TYPE=GEGR, LOCXYZ, CH RALAR) INSTANCES IN KM, ANGLES IN DEG)	0. 105. 36. 0.	NC NO NO NO
TGPOS	4	TGPOS(1-4) = TARGET (IN IMPACT) POSITION (TGPS(1-4)=TARGET CCOC, TGCS(4)=ECCH TYPE=GEGR, LCCXZ, CH PACAK) (INSTANCES IN KM, ANGLES IN DEG)	0. 0. 0. 0.	GEOPP 0. NO NO
GAMA	1	GAMA = HEELING ANGLE FOR TRAJECTORY SPECIFICATION (DEFAULT UNIT IS RADIANS)	0.	NO
TEMP	1	TEMP = IMPACT TIME FOR 1-ST HY (REFALY UNIT IS SEC)	20.	CEG YES
TOELT	1	TOELT = DELTA TIME BETWEEN HVS (DEFAULT UNIT IS SEC)	200.	SEC YES
		- SATEC INPUTS		
		-- TO RUN A SATEC PROBLEM, INPUT THE FIRST SATEC CALCULATION TIME (CTIME) TO CCUR PRIOR TO THE PROBLEM STOP TIME (TSTOP).		
		1. RUN CONTROL		

DICTIONARY OF INPUT VARIABLES

Table A.1 (Continued)

INPUT
NAMEVARIABLE
VALUES

DESCRIPTION

02/04/80 10.36.18.

PAGE 7

DEFAULT
VALUESUNIT-NAME
ALLOED--SUBSEQUENT SATCOM CALCULATIONS ARE PERFORMED EVERY
(CTINT) SEC.

CTIME 1 CTINT = FIRST SATCOM CALCULATION TIME (DEFAULT UNIT IS SEC)

CTINT 1 CTINT = TIME STEP FOR SAT-COM CALCULATIONS (DEFAULT UNIT IS SEC)

2. PROCESSING DATA

--FOR A MORE DETAILED DESCRIPTION OF THESE INPUTS SEE
THE RESSOC MANUAL VOL. 20.

CTYPE	1	CTYPE = SATCOM MODULATION TYPE (SAPACFSK, CR FSK)	CPSK	NO
REGEN	1	REGEN = FLAG FOR REGENERATION OF SIGNAL AT SATELLITE (YES OR NO)	YES	NO
COMIT	1	COMIT = FLAG FOR CURRENT FSK MODULATION (YES OR NO)	NO	NO
DETRM	1	DETRM = FLAG FOR FULLY DETERMINISTIC MODE (CALCULATIONS (YES OR NO)	NO	NO
ORDER	1	ORDER = ORDER OF PHASE LOOKUP (FIRST OR SECOND)	YES	NO

3. PLATFORM DATA

--RELATIVE COORDINATES CAN BE USED HERE TO ALIGN THE
COMMUNICATIONS LINKS AND EARTH REGIONS.

XPOS	4	XPOS(1-4) = TRANSmitter POSITION (XPOS(1-3)=POSITION CCGRC, YPOS(4)= CCGRC TYPE--GECGH,LCXYZ,CR FACAH) (CISTANCES IN KM, ANGLES IN DEG)	0.	0.
------	---	---	----	----

RPOS	4	RPOS(1-4) = RECEIVER POSITION (RPOS(1-3)=POSITION CCGRC, RPOS(4)= CCGRC TYPE--GECGH,LCXYZ, CR FACAH) (CISTANCES IN KM, ANGLES IN DEG)	0.	0.
------	---	---	----	----

SPOS	4	SPOS(1-4) = SATELLITE POSITION (SPOS(1-3)=POSITION COORDN. SECOS(4)= CCGRC TYPE--GECGH,LCXYZ,CR FACAH) (CISTANCES IN KM, ANGLES IN DEG)	0.	0.
------	---	--	----	----

		35767.	35767.	LOCXYZ
--	--	--------	--------	--------

4. LINK DATA

POWER	2	POWER(1-2) = TRANSMITTER POWER (UPLINK, DOWNLINK) (DEFAULT UNIT IS WATTS)	100.	WATTS
CFREQ	2	CFREQ(1-2) = SATCOM FREQUENCY (UPLINK, DOWNLINK) (DEFAULT UNIT IS MHZ)	20.	YES
XGAIN	2	XGAIN(1-2) = TRANSMITTER GAIN (UPLINK, DOWNLINK) (DEFAULT UNIT IS RATIO (0.1E-SIGNALLESS))	8000.	YES
RGAIN	2	RGAIN(1-2) = RECEIVER GAIN (UPLINK, DOWNLINK) (DEFAULT UNIT IS RATIO (0.1E-SIGNALLESS))	7400.	YES

		61.	0.8	YES
--	--	-----	-----	-----

		33.2	0.8	YES
--	--	------	-----	-----

		16.0	0.8	YES
--	--	------	-----	-----

		61.	0.8	YES
--	--	-----	-----	-----

DIRECTORY OF INPUT VARIABLES

Table A.1 (Continued)
INPUT
NO.
VARIABLE
VALUES

		DESCRIPTION	DEFAULT VALUES	INIT-NAME ALLOWED
BITP	2	BITP(1-2) = BIT PITCH (UPLINK, LOCALLINK) (DEFAULT UNIT IS SEC)	1.0E-8	SEC
CBAND	2	CBANC(1-2) = IF FILTER BANDWIDTH (UPLINK, CCNLINK, CCNLINK) (DEFAULT UNIT IS Hz)	1.0E-8	SEC
PBAND	2	PBAND(1-2) = PLL BANDWIDTH (UPLINK, CCNLINK) (DEFAULT UNIT IS Hz)	125.	MHZ
CBEAM	2	CBEAM(1-2) = TRANSMITTER BEAMWIDTH (UPLINK, DOWNLINK) (DEFAULT UNIT IS RADIANS)	125.	MHZ
CSN	2	CSN(1-2) = SATCOM S/N THRESHOLD (UPLINK, CCNLINK) (DEFAULT UNIT IS RATIO (UNITSLESS))	15.	DB

E. OPTICS CODE INPUTS

**TO RUN ANY OPTICS PROBLEM THE USER MUST FIRST SET
OPTICS=REFIN

**TWO TYPES OF OPTICS PROBLEMS CAN BE SIMULATED--

(1) A SURVEILLANCE PROFILE WHERE THE STREAM IS POINTED
AT SOME REFERENCE LOCATION (UTPESURVEYFILE), OR
(2) A EARTH TRACK PROFILE WHERE A STREAM LOCK IS
CREATED INTERNALLY AT A SPECIFIED TIME (UTPE=TRACK).

--IN THE FIRST CASE, THE USER PROVIDES THE FIRST LOCK
TIME(LOC0) AND THE REFERENCE POINT (REFP0) FOR
THE LOCK DIRECTION (OR TRACK THE BURST 1 FIREBALL)
(SEE SERVO INPUT). IN THE SECOND CASE, THE USER INPUTS
THE BCASTER PCSEL AND THE PULLOUT POSITION (BGPOS).
IN EITHER CASE, THE USER MUST SET UP AN OPTICAL SENSOR

1. RUN COPTIC

--SUBSEQUENT OPTICS LOCKS ARE CREATED INTERNALLY EVERY
(IFTIME) SECONDS.

--OPTICS OUTPUTS ARE CONTROLLED BY THE (OPTCL) PARAMETER
(1) FOR LOCALCPWINTS). BOOST TRACK MEASUREMENTS ONLY
ARE PROVIDED, AND
(2) FOR LOCALCPW). DATA STREAM OUTPUT AS THE DETECTOR

SCANS THE FLY AND ALSO BE PRODUCED.

--THE ECGSTEN MEASUREMENTS MAY BE USED TO INITIALIZE
(IN ACC TO) A TRACK FILE BY SETTING (TFILE=REFER), AND
THESE MEASUREMENTS MAY BE NFTIED WITH RADAR DATA BY
SETTING (SNL=TEST).

OPTICS	1	OPTICS = FLAG FOR INITIALIZING OPTICS CALCULATION (SET OPTICS=REFER ZERUS NO
--------	---	--

DIRECTORY OF INPUT VARIABLES

Table A.1 (Continued)

INPUT VARIABLE	NO. VALUES	DESCRIPTION
OVTYP	1	OVTYP = OPTICS LUM TYPE (TRACK OR SURVEILANCE) (FOR CITY-SURVEILANCE) (DEFAULT UNIT IS SEC)
OLOOK	1	CLOCK = TIME OF FIRST OPTICS LOOK (FOR CITY-SURVEILANCE) (DEFAULT UNIT IS SEC)
FTIME	1	FTIME = FRAME TIME FOR OPTICS LOOKS (DEFAULT UNIT IS SEC)
OCAUC	1	OCALC = OPTICS CALCULATION TYPE (PCNTS ON FOV)
TFILE	1	TFILE = OPTICAL TRACK FILE FLAG (TRUE-FALSE, FOR TRACK FILE, --ZEROS FOR IC TRACK FILE)
SNET	1	SNET = SENSOR NETTING FLAG (TRUE OR FALSE)
SENSPT	1	SENSPT = TYPE OF TARGET SENSOR IS PLACED ICARD. USE HEF FOR A FIXED POINT (HEFP1). FIREBALL TO TRACK THE FIREBALL OF BLASTS. NOTE THE LSIL MUST ALSO SET REFER=REFER AND CBTAGREF=OBJECT-- SEE SECTION C-3 ABOVE.
REFPT(1-4)	4	REFPT(1-4) = REFERENCE POINT FOR SENSOR POSITION (HEFP1-3)=POSITION COORD,REFPT4=CCNU. TYPE -- GEGR, LOCXY, ON RADAR, (INSTANCES IN LEG) (ANGLES IN RAD, ANGLES IN DEG)

2. SENSOR DATA

--THE SENSOR LOCATION CAN BE INPUT IN GEOGRAPHICAL COORDINATES (GEGR) OR RELATIVE TO THE REFERENCE LOCATION IN SECTION A. ABOVE.
--THERE ARE TWO AVERAGING BANDS ALLOCATED AND THREE BUILT IN SURVEILANCE WHEELS. THE FIRST TWO WHEELS SHOULD BE USED IN SURVEILANCE APPLICATIONS AND PROVIDE SLIGHTLY DIFFERENT CUTOFF. THE THIRD WHEEL (SURVEIL-04) PROVIDES TRACK MEASUREMENT OUTPUT AND SHOULD BE USED WHEN (TFILE=HEFLH).

SNPOS	4	FOR OPTICAL SENSOR POSITION (SNPOS1-3)=POSITION, COORD, ANGLES IN DEG	357.87, -79.33, 47.75, NO, NO, NO
MLC(1-2)	2	MLC(1-2) = LOW END OF SURVEIL WAVELENGTH BAND -- (TWO BANDS ALLOWED)	2.5L-6, M, YES, NO
MLI(1-2)	2	MLI(1-2) = HIGH END OF SURVEIL WAVELENGTH BAND (DEFAULT UNIT IS CM)	2.6L-6, M, YES, YES
OFLHR	2	OFLHR(1-2) = FIXE POSITION OF OPTICS RADAR MEASUREMENT EHRCRS IN AE COORD (DEFAULT UNIT IS RADIAN)	2.7E-6, M, YES, YES
OSMER(1-2)	2	OSMER(1-2) = S/N DEPENDENT POSITION OF OPTICS RADAR MEASUREMENT EHRS IN AE CCNU (DEFAULT UNIT IS RADIAN)	.01, MRAO, YES, YES
ONRCL	1	ONRCL = OPTICAL SENSOR PROCESSING MODEL -- (SURVEIL-01, SURVEIL-02, OR SURVEIL-04)	1, MRAO, YES, NO

3. BOOSTER DATA

--TWO BOOSTER STAGES ARE ALLOWED. NOTE THAT THE TIME CORRESP. TO THE INITIAL BOOSTER STATE IS SET INTERNALLY TO 0. SEC AND THE AV IMPACT TIME SPECIFIED IN THE

DIRECTORY OF INPUT VARIABLES
Table A.1 (Continued)
INPUT
NO.
VARIABLE
VALUES

		DESCRIPTION	DEFAULT VALUES	INPUT-PAIR ALLOWED
THRETEEN INPUTS (SEE SECTION C.3) WILL BE ADJUSTED ACCORDINGLY.				
FUEL	2	FUEL(1-2) = FUEL TYPE (LIGUID OR SOLID)--NOTE--TWO STAGES ALLOWED	LIGUID	NO
THRST	2	THRST(1-2) = BOOSTER STAGE THRUST (DEFAULT UNIT IS GM)	110000.	LC
ATI	2	ATI(1-2) = INITIAL STAGE WEIGHT (DEFAULT UNIT IS GM)	135000.	YES
ATF	2	ATF(1-2) = FINAL STAGE WEIGHT (DEFAULT UNIT IS GM)	70000.	YES
AN02	2	AN02(1-2) = NOZZLE EXIT AREA (DEFAULT UNIT IS CMSQ)	20000.	LB
TRUN	2	TRUN(1-2) = STAGE BURN TIME (DEFAULT UNIT IS SEC)	35000.	YES
REFA	2	REFA(1-2) = REFERENCE AREA FOR AERODYNAMIC CRAG CALCULATION (DEFAULT UNIT IS CMSQ)	8000.	LB
CX0	2	CX0(1-2) = AXIAL FORCE COEFFICIENT AT M=0.5	3000.	INSA
CX1	2	CX1(1-2) = AXIAL FORCE COEFFICIENT FOR M>1.0	2000.	INSC
CX2	2	CX2(1-2) = AXIAL FORCE COEFFICIENT FOR M>3.0	0.	SEC
				YES
				FTSQ
				35.
				FTSQ
				NO

DESCRIPTION OF USER INPUT AND COMMAND FORMATS . . .

Table A.1 (Concluded.)

THE BASIC FORM FOR EACH INPUT LINE IS . . .

11.12.13.

(ALL BLANKS ON THE LINE ARE IGNORED)
WHERE THE 11, 12, ETC. ARE EITHER COMMANDS OR ITEMS OF THE FORM . . .

ITEM=LIST

WHERE ITEM IS ONE OF THE INPUT VARIABLE, OR VECTOR ELEMENTS AND
LIST IS A LIST OF ONE OR MORE VALUES TO BE INPUT, STARTING AT THE
NAMED ELEMENT. THE VALUES NEED NOT INCLUDE DECIMAL POINTS FOR
WHOLE NUMBERS AND MAY BE APPENDED WITH APPROPRIATE UNIT NAMES IF
ALLOWED FOR THAT VARIABLE. VALUES ARE SEPARATED BY COMMAS.

THE RECOGNIZED COMMANDS ARE . . .

ABORT	CAUSES PROGRAM ABORT WITH NO OUTPUT FILE (TC AVAIL SUBMITTING A BATCH JOB)
CHANGELISTON	Turns on substitution list option (shows how values are used in ROSCOE input deck)
CHANGELISTOFF	Turns change list option off
HELP	To process this menu again
RUN	Terminates execution and processes output file for ROSCOE execution. ALTERNATE FORMS ARE END OR END DATA

TABLE A.2

ALLOWABLE UNIT NAMES

<u>Category</u>	<u>Unit Name</u>	<u>Scaling Factor to Internal (Default) Units</u>
Frequency	MHZ	1,000,000
	KHZ	1,000
Time	HRS	1 (This may only be used for time-of-day inputs)
	SEC	1
Mass	KG	1,000
	GM	1
	LB	453.592
Ballistic Coeff.	PSF	0.4882405
	GM/CMSQ	1
Length	CM	1
	FT	30.48
	KM	100,000
	NMI or NM	185,325
	M	100
	KFT	30,480
Acceleration	G	980.665
Area	CMSQ	1
	MSQ	10,000
	INSQ	6.4516
	FTSQ	929.0304

TABLE A.2 (Cont'd.)

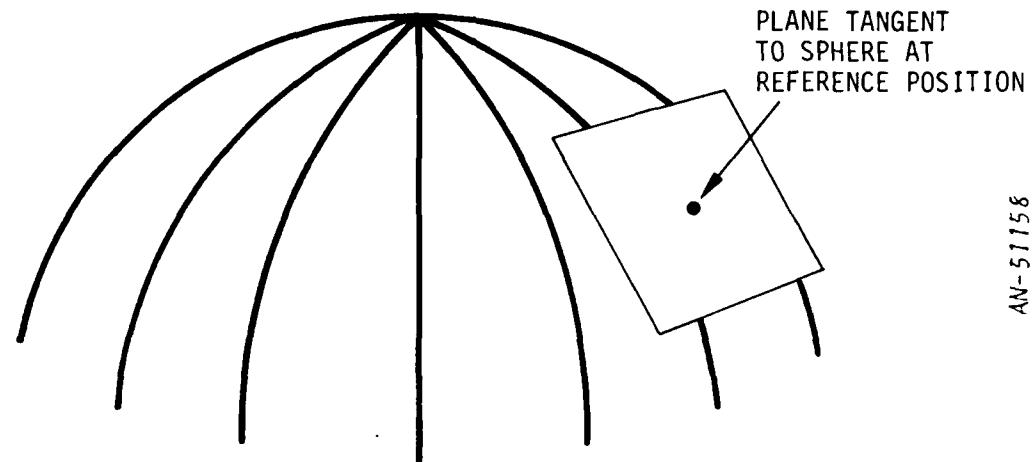
ALLOWABLE UNIT NAMES

<u>Category</u>	<u>Unit Name</u>	<u>Scaling Factor to Internal (Default) Units</u>
Yield	MT	1
	KT	0.001
Radar Range/Standard Target	CMSQCM	1
	KMSQM	10,000
	NMSQM	18532.5
	KFSQM	3048
Power	WATTS	10,000,000
Power Ratio	DB	$\times 10^{dB/10}$
Angle	DEG	0.01745329252
	RAD	1
	MRAD	0.001

TABLE A.3

POSITION COORDINATE SPECIFICATIONS

GEOGR	Geographical Coordinates: <ul style="list-style-type: none">• Altitude (KM)• East longitude (DEG) (longitudes west of Greenwich input as negative)• North latitude (DEG) (south latitudes negative)
LOCXYZ	Local Tangent Plane Coordinates (see Fig. A.1): <ul style="list-style-type: none">• Geographic east (KM) (west input as negative)• Geographic north (KM) (south input as negative)• Distance above plane (KM)
RADAR	Local Radar Coordinates (see Fig. A.1); <ul style="list-style-type: none">• Slant range (KM)• Azimuth (DEG) (positive CCW from east)• Elevation (DEG) (positive above horizontal)



AN-51156

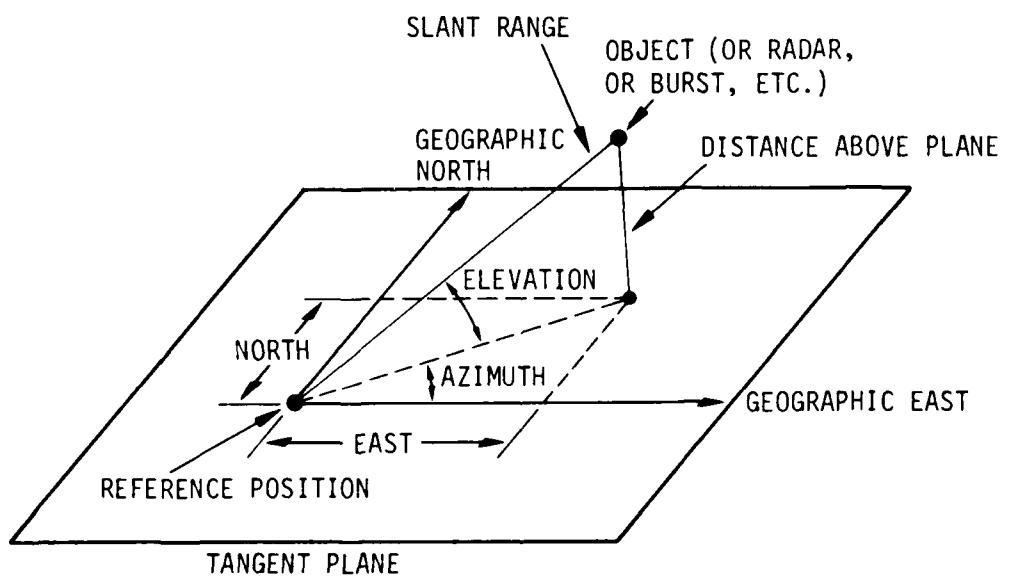


Figure A.1 Definition of Coordinates for Relative Coordinate Systems

TABLE A.4
SAMPLE CONTROL CARD DECK FOR AFWL/NOS/BE1

```

JOB CARD * * * *
ACCOUNT CARD * * * *
MAP(OFF)
ATTACH(XX1,0BINARY, ID=GRCXJUE,CY=1)
COPYBK(XX1,0BIN,240)
ATTACH(XX2,0BINARY, ID=GRCXJUB,CY=2)
COPYSF(XX2,0BIN)
RETURN(XX1,XX2)
REWIND(OBIN)
ATTACH(BCPYL,0CPYLROSCL, ID=GRCXJUB,CY=3)
ATTACH(STRUCT,STRUCT, ID=GRCXJUB)
UPDATE(P=STRUCT,F,D,B,C=TAPL1,L=1)
BCPYL(TAPL1,0BIN,LFILE,,REAL1,REWIND,ERRORS)
RETURN(TAPE1,TAPE4,BCPYL,0BIN)
ATTACH(SPIRE,SPINEROSCL, ID=GRCXJUB,CY=2,MR=1)
ATTACH(TAPE1,TAPL1,SPINEROSCL, ID=GRCXJUB,CY=3)
SPINL(,,TAPL1,1,,ATL,REWIND)
ATTACH(RLIBE,RLIBEROSCL, ID=GRCXJUB)
RETURN(TAPE1,TAPE2,TAPE3,TAPE4,TAPL5,TAPE6)
ATTACH(ANALGM8,ANALGM8ROSCL, ID=GRCXJUB)
ANALGM8.
RETURN(ANALGM8)
LOSET(LIB=RLIBE,PRESET=ZERO,FILES=TAPL1)
LOAD(LFILE)
NOGO.
RETURN(LFILE)
RETURN(RLIBE)
ATTACH(TAPE3,NEWCATROSCL, ID=GRCXJUB)
SENDER.
7-8-9 CARD
*IDENT QCHG
*COMPILE STRUCT
ANY MOUS TO STRUCT FILE GO HERE * * * *
7-8-9 CARD
CHANGE LIST CN * * * * * (OPTIONAL)
SPINL DATA INPUTS * * * *
* * * *
* * * *
RJN
7-8-9 CARD
6-7-8-9 CARD

```

TABLE A.5

SAMPLE PROCEDURE PERMFILE FOR INTERACTIVE USE

```

.PROC,ROSCOTS
COPYCR(ROSCOTS,DATA01K,2)
ATTACH(SPIRE,SHINEROSCOL,1D=GRCXJJB,CY=2,MR=1)
ATTACH(CTTAB,DATA1EROSCOL,1D=GRCXJJB,CY=5)
SPIRE.
RETURN(SPIRE,INTAB,WAFILE)
ZAP(INTAB,WW,INT)
COMMENT. FILE HAS BEEN BATCHED TO INPUT.
7-8-9 CARD
JOB CARD * * * *
ACCOUNT CARD * * * *
MAP(OFF)
ATTACH(XX1,CBINARY,1D=GRCXJJB,CY=1)
COPYBR(XX1,CB1N,<40)
ATTACH(XX2,CBINARY,1D=GRCXJJB,CY=2)
COPYBF(XX2,CB1N)
RETURN(XX1,XX2)
REWIND(GLIN)
ATTACH(FCPYL,FCPYLEROSCOL,1D=GRCXJJB,CY=3)
ATTACH(STRUCT,USTRUCT,1D=GRCXJJB)
UPDATE(P=STRUCT,F=0,8,L=TAPE1,L=1)
BCPYI(TAPE1,08IN,LF1LE,,READ1,REWIND,ERRORS)
RETURN(TAPE1,TAPE4,BCPYL,08IN)
COPYCR(INPUT,INDATA)
REWIND(INDATA)
ATTACH(RLIBE,RLIBEROSCOL,1D=GRCXJJB)
RETURN(TAPE1,TAPE2,TAPE3,TAPE4,TAPE5,TAPE6)
ATTACH(AMALGM8,AMALGM8EROSCOL,1D=GRCXJJB)
AMALGM8.
RETURN(AMALGM8)
LOAD(LFILE)
NOGU.
RETURN(LFILE)
RETURN(RLIBE)
ATTACH(TAPES,NEWLATROSCOL,1D=GRCXJJB)
SENSEL.
7-8-9 CARD
*IDENT SONG
*COMPILE STRUCT
ANY MOUS TO OSSTRUCT FILE GO HERE * * * *
7-8-9 CARD
6-7-8-9 CARD

```

TABLE A.6
TIME-SHARE INPUTS
(Underlined portions typed by User)

1. COMMAND - ATTACH (ROSCOTS, ID = GRCXJJB)

2. COMMAND - RØSCØTS

3. INPUTS? (USER TYPES IN INPUTS)

INPUTS? (USER TYPES IN INPUTS)

· · ·
· · ·
· · ·

ERRORS - (--IF THERE ARE INPUT ERRORS, RØSCØTS LISTS
THEM HERE AND REQUESTS INPUTS AGAIN)]

INPUTS? RUN

DISTRIBUTION LIST

DEPARTMENT OF DEFENSE

Assistant to the Secretary of Defense
Atomic Energy
ATTN: Executive Assistant

Defense Advanced Rsch Proj Agency
ATTN: STO, S. Zakanycz

Defense Communications Engineer Center
ATTN: Code R410, J. McLean

Defense Nuclear Agency
ATTN: RAAE, P. Lunn
ATTN: RAAE
4 cy ATTN: TITL

Defense Technical Information Center
12 cy ATTN: DD

Field Command
Defense Nuclear Agency
ATTN: FCPR

Field Command
Defense Nuclear Agency
Livermore Branch
ATTN: FCPR

National Security Agency
ATTN: R-52, J. Skillman

Under Secretary of Def for Rsch & Engrg
ATTN: Strategic & Space Sys (OS)

WWMCCS System Engineering Org
ATTN: R. Crawford

DEPARTMENT OF THE ARMY

Atmospheric Sciences Laboratory
U.S. Army Electronics R&D Command
ATTN: DELAS-E0, F. Niles

BMD Advanced Technology Center
Department of the Army
ATTN: ATC-T, M. Capps
ATTN: ATC-O, W. Davies

BMD Systems Command
Department of the Army
ATTN: BMDSC-HW, R. Dekalb

Harry Diamond Laboratories
Department of the Army
ATTN: DELHD-N-P, F. Wimenitz
ATTN: DELHD-I-TL

U.S. Army Foreign Science & Tech Ctr
ATTN: DRXST-SD

U.S. Army Missile Intelligence Agency
ATTN: J. Gamble

U.S. Army Missile R&D Command
ATTN: DRDMI-XS
ATTN: RSIC

DEPARTMENT OF THE ARMY (Continued)

U.S. Army Nuclear & Chemical Agency
ATTN: Library

U.S. Army Satellite Comm Agency
ATTN: Document Control

U.S. Army TRADOC Sys Analysis Actvty
ATTN: ATAA-PL

DEPARTMENT OF THE NAVY

Naval Electronic Systems Command
ATTN: PME 117-20
ATTN: Code 501A

Naval Intelligence Support Ctr
ATTN: Document Control

Naval Ocean Systems Center
ATTN: Code 532

Naval Postgraduate School
ATTN: Code 1424, Library

Naval Research Laboratory
ATTN: Code 2627
ATTN: Code 4709, W. Ali
ATTN: Code 4701, J. Brown
ATTN: Code 4780, S. Ossakow
ATTN: Code 4700, T. Coffey
ATTN: Code 4780, P. Palmadesso

Naval Surface Weapons Center
ATTN: Code X211

Strategic Systems Project Office
Department of the Navy
ATTN: NSP-2722, F. Wimberly
ATTN: NSSP-2722, M. Mesarole

DEPARTMENT OF THE AIR FORCE

Air Force Geophysics Laboratory
ATTN: OPR, A. Stair
ATTN: SULL
ATTN: LKB, K. Champion
ATTN: OPR, H. Gardiner

Air Force Systems Command
ATTN: Technical Library

Air Force Technical Applications Ctr
ATTN: TFR, C. Meneely
ATTN: Technical Library

Air Force Weapons Laboratory
Air Force Systems Command
ATTN: NYC
ATTN: SUL

Air University Library
Department of the Air Force
ATTN: AUL-LSE

DEPARTMENT OF THE AIR FORCE (Continued)

Ballistic Missile Office
Air Force Systems Command
ATTN: MNRT
ATTN: MNX
ATTN: MNRC

Deputy Chief of Staff
Research, Development, & Acq
Department of the Air Force
ATTN: AFROS

Headquarters Space Division
Air Force Systems Command
ATTN: SKX
ATTN: SKA, M. Clavin

Headquarters Space Division
Air Force Systems Command
ATTN: SZJ, P. Kelley

Rome Air Development Center
Air Force Systems Command
ATTN: OCSA, J. Simons
ATTN: OCS, V. Coyne
ATTN: TSLD

Strategic Air Command
Department of the Air Force
ATTN: NRT
ATTN: XPFS, B. Stephan

DEPARTMENT OF ENERGY

Department of Energy
ATTN: OMA

OTHER GOVERNMENT AGENCIES

Department of Commerce
National Oceanic & Atmospheric Admin
ATTN: F. Fehsenfeld

Institute for Telecommunications Sciences
National Telecommunications & Info Admin
ATTN: G. Falcon
ATTN: W. Utlaut

DEPARTMENT OF DEFENSE CONTRACTORS

Aerojet Electro-Systems Co
ATTN: J. Graham

Aerospace Corp
ATTN: J. Strauss
ATTN: J. Reinheimer
ATTN: N. Stockwell
ATTN: I. Garfunkel
ATTN: N. Cohen
ATTN: V. Josephson

Berkeley Research Associates, Inc
ATTN: J. Workman

ESI, Inc
ATTN: J. Marshall

General Electric Co
ATTN: M. Bortner

DEPARTMENT OF DEFENSE CONTRACTORS (Continued)

General Research Corp
ATTN: J. Ise, Jr
ATTN: J. Garbarino

Jamieson Science & Engineering
ATTN: J. Jamieson

Kaman Sciences Corp
ATTN: D. Perio
ATTN: N. Beauchamp
ATTN: P. Tracy

Kaman TEMPO
ATTN: W. Knapp
ATTN: K. Schwartzman
ATTN: M. Stanton
ATTN: T. Stephens
ATTN: J. Jordano
ATTN: DASIAC

Lockheed Missiles & Space Co, Inc
ATTN: D. Divis

Lockheed Missiles & Space Co, Inc
ATTN: M. Walt

M.I.T. Lincoln Lab
ATTN: D. Towle

McDonnell Douglas Corp
ATTN: H. Spitzer
ATTN: R. Halprin

Mission Research Corp
ATTN: R. Kilb
ATTN: R. Hendrick
ATTN: M. Scheibe
ATTN: D. Sappenfield
ATTN: D. Archer
ATTN: R. Bogusch
ATTN: F. Fajen

Nichols Research Corp, Inc
ATTN: N. Byrn

Pacific-Sierra Research Corp
ATTN: H. Brode

Photometrics, Inc
ATTN: I. Kofsky

Physical Research, Inc
ATTN: R. Deliberis

University of Pittsburgh
ATTN: F. Kaufman

R & D Associates
ATTN: R. Lelevier
ATTN: F. Gilmore
ATTN: B. Gabbard
ATTN: R. Turco
ATTN: P. Haas

R & D Associates
ATTN: B. Yoon

DEPARTMENT OF DEFENSE CONTRACTORS (Continued)

Rand Corp
ATTN: C. Crain

Science Applications, Inc
ATTN: D. Hamlin

Science Applications, Inc
ATTN: W. Mendes

SRI International
ATTN: W. Chesnut
ATTN: W. Jaye

DEPARTMENT OF DEFENSE CONTRACTORS (Continued)

Teledyne Brown Engineering
ATTN: J. Cato
ATTN: G. Harney
ATTN: Technical Library
ATTN: J. Ford

Visidyne, Inc
ATTN: H. Smith
ATTN: C. Humphrey
ATTN: J. Carpenter